

SUPPLEMENTAL TYPE CERTIFICATE

10017072 REV. 1

This Supplemental Type Certificate is issued by EASA, acting in accordance with Regulation (EC) No. 216/2008 on behalf of the European Community, its Member States and of the European third countries that participate in the activities of EASA under Article 66 of that Regulation and in accordance with Commission Regulation (EU) No. 748/2012 to

DONALDSON COMPANY, Inc.

**17891 CHESTERFIELD AIRPORT ROAD
CHESTERFIELD MO 63005
USA**

and certifies that the change in the type design for the product listed below with the limitations and conditions specified meets the applicable Type Certification Basis and environmental protection requirements when operated within the conditions and limitations specified below:

Original Type Certificate Number: EASA.R.008

Type Certificate Holder: AIRBUS HELICOPTERS

Type: AS 350/EC 130

Model: AS350 B, AS350 B1, AS350 B2

AS350 B3, AS350 BA

Original STC Number: FAA STC SR02393CH

Description of Design Change:

Installation of Inlet Barrier Filter system

Revision 1 reflects the name and address change from Aerospace Filtration Systems, Inc. to Donaldson Company, Inc.

EASA Certification Basis:

The Certification Basis (CB) for the original product remains applicable to this certificate/ approval.

The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.


Associated Technical Documentation:

1. Master Data List No. AFS-AS350B-IBF-MDL Revision F dated 5 May 2008 or later EASA approved revision.

See Continuation Sheet(s)

For the European Aviation Safety Agency

Cologne, Germany, 31 January 2018



Pier Giorgio COLOMBO
Medium Rotorcraft Section
Manager

2. Rotorcraft Flight Manual Supplement No. AFS-AS350B-IBF-FMS Revision B dated 8 Sep. 2008 or later EASA approved revision.

3. Instructions for Continued Airworthiness No. AFS-AS350B-IBF-ICA Revision B dated 5 May 2007 or later EASA approved revision.

Limitations/Conditions:

Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product.

- End -