

SUPPLEMENTAL TYPE CERTIFICATE

10026895 REV. 1

This Supplemental Type Certificate is issued by EASA, acting in accordance with Regulation (EC) No. 216/2008 on behalf of the European Community, its Member States and of the European third countries that participate in the activities of EASA under Article 66 of that Regulation and in accordance with Commission Regulation (EU) No. 748/2012 to

DONALDSON COMPANY, Inc.

**17891 CHESTERFIELD AIRPORT ROAD
CHESTERFIELD MO 63005
USA**

and certifies that the change in the type design for the product listed below with the limitations and conditions specified meets the applicable Type Certification Basis and environmental protection requirements when operated within the conditions and limitations specified below:

Original Type Certificate Number: US H19NM

Type Certificate Holder: MD HELICOPTERS INC.

Type: MD900

Model: MD900

Original STC Number: FAA STC SR02526CH

Description of Design Change:

Installation of engine inlet barrier filter (IBF) system

Revision 1 reflects the name and address change from Aerospace Filtration Systems, Inc. to Donaldson Company, Inc.

EASA Certification Basis:

The Certification Basis (CB) for the original product remains applicable to this certificate/ approval.

The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.

Associated Technical Documentation:

1. Master Data List Inlet Barrier Filter system AFS-MD900-IBF-MDL Rev D dated 15 July 2009.

See Continuation Sheet(s)

For the European Aviation Safety Agency

Cologne, Germany, 31 January 2018



Pier Giorgio COLOMBO
Medium Rotorcraft Section
Manager



2. Aircraft Flight Manual Supplement AFS-MD900-IBF-FMS Rev A approved by the FAA on 21 July 2009.

3. Instructions for Continued Airworthiness AFS-MD900-IBF-ICA Rev IR dated 20 October 2006.

or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision)" and/ or the Technical Implementation Procedures of EU/ USA Bilateral Agreement.

Limitations/Conditions:

Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product.

- End -